

# A New Multiobjective Genetic Programming for Extraction of Design Information from Non-dominated Solutions

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# Background

## Multi-Objective Design Exploration(MODE)

*Obayashi, et.al. , "Multi-Objective Design Exploration for Aerodynamic Configurations" , 2005*

Problem Setting



Multi-objective optimization



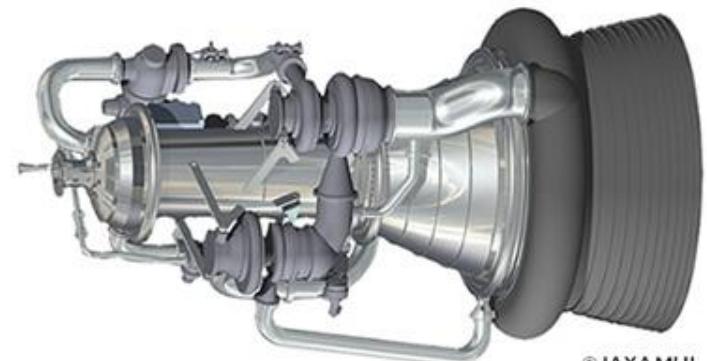
Data mining methods



Extraction of design  
knowledge



Mitsubishi Regional Jet (MRJ)



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LE-X

# Multi-Objective Design Exploration

## - Problem setting -

### ■ Design parameters

Kink position ( $x, y$ )

### ■ Objective functions

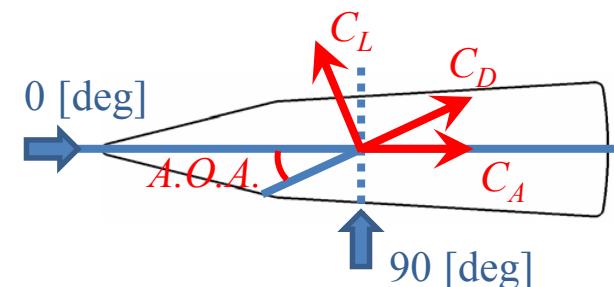
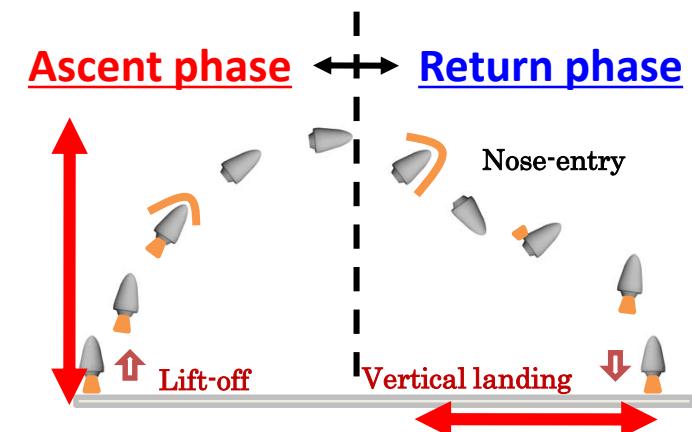
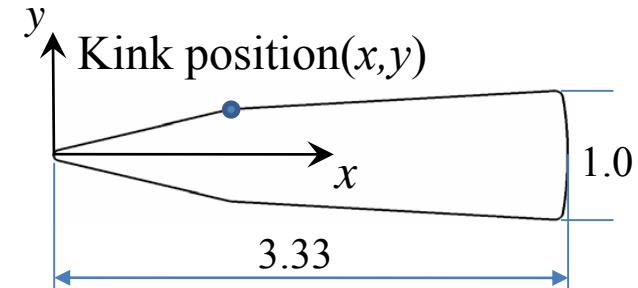
#### Ascent phase

1. Drag to be minimized ( $M=2.0$ ,  $AOA = 0$ )

#### Return phase

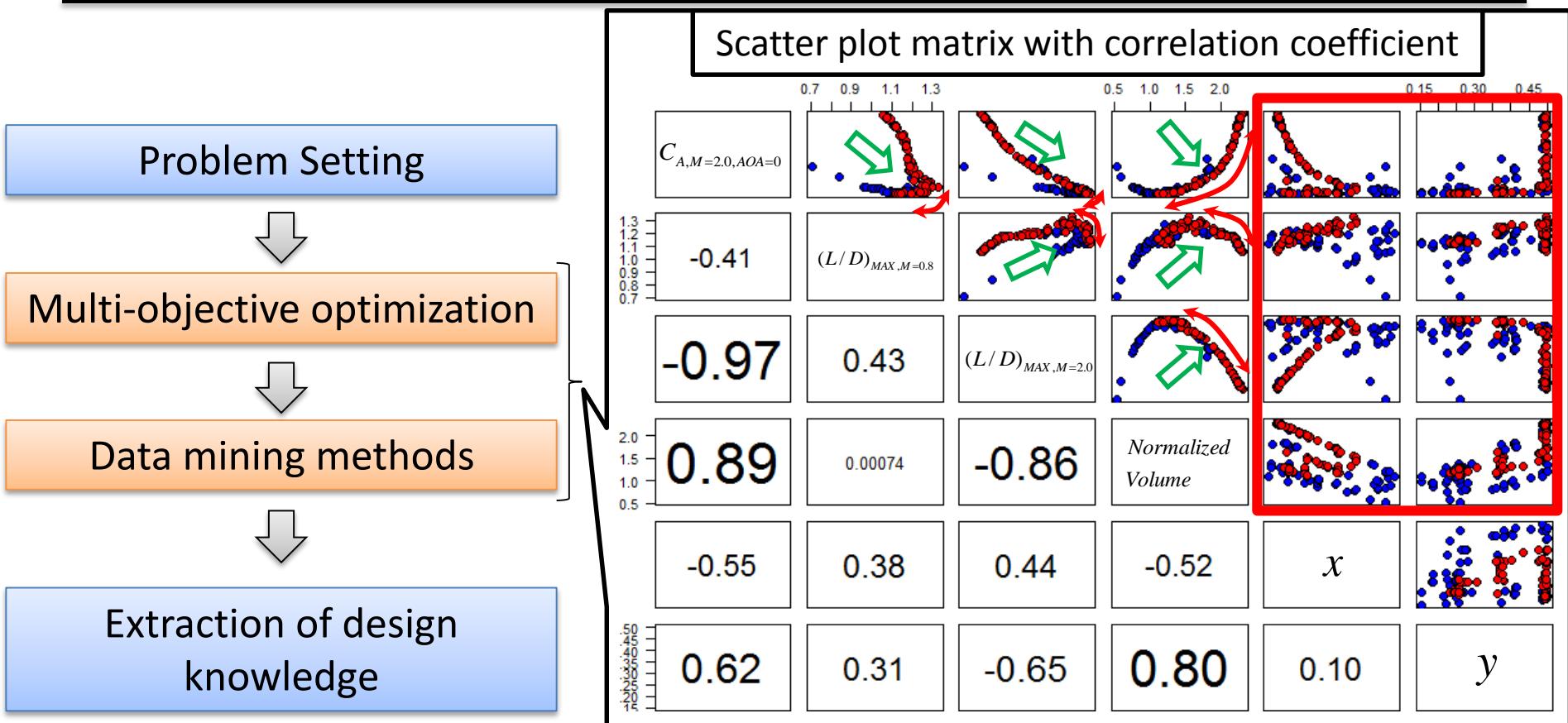
2. Maximum L/D to be maximized at subsonic condition ( $M=0.8$ )
3. Maximum L/D to be maximized at supersonic condition ( $M=2.0$ )

4. Body volume to be maximized



# Multi-Objective Design Exploration

- Multi-objective optimization & Data mining -



- It is difficult to reveal non-linear relationship ( $\sin(x_1)$ ,  $x_1x_2$ ,  $x_1/x_2$ ) between parameters

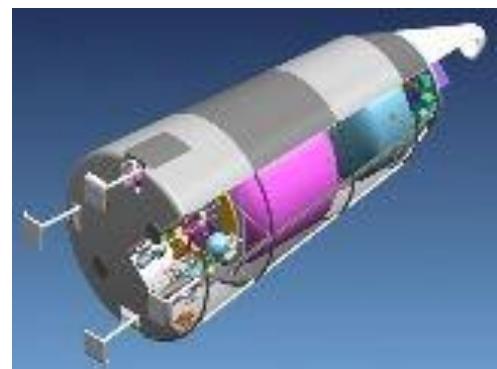
→ Genetic programming is one of the evolutionary technique to reveal non-linear relationship.

# Objective

- Present a new data mining method for MODE to extract design information from non-dominated solutions

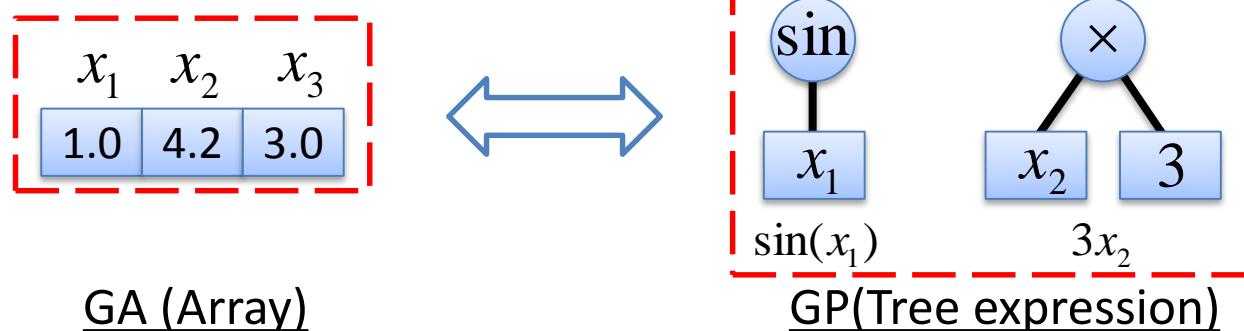
## Approach

- Multi-objective GP(MOGP) for MODE is introduced
- MOGP is applied to the practical multi-objective optimization problem of reusable launch vehicle

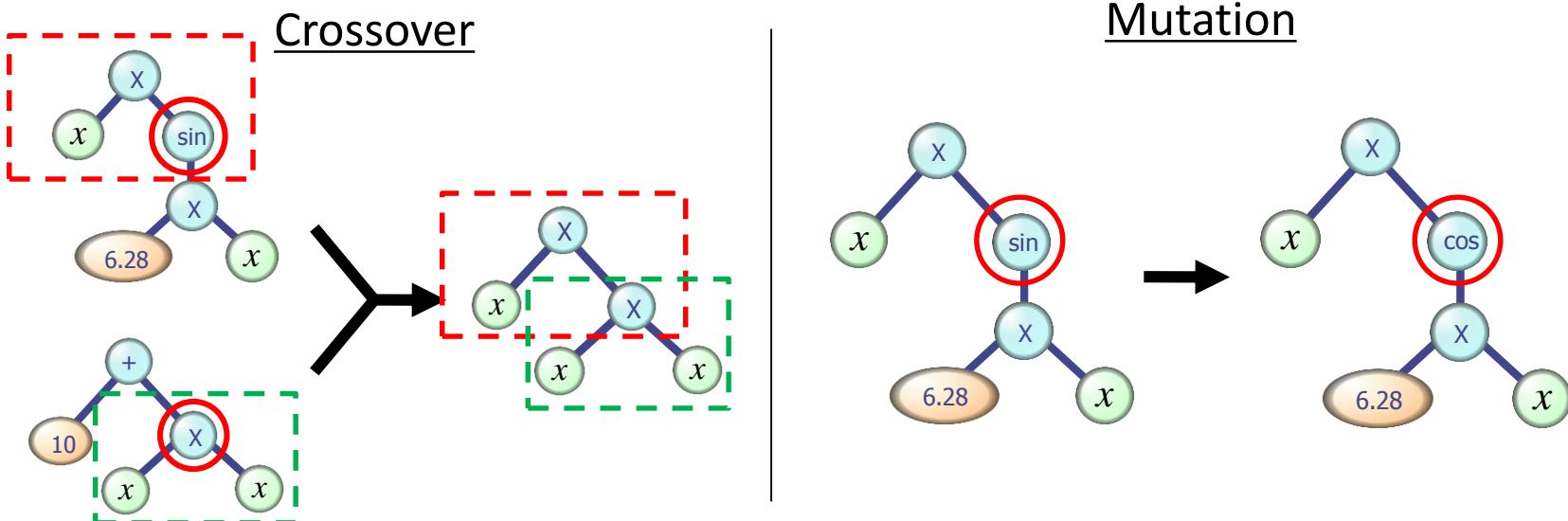


# About Genetic Programming

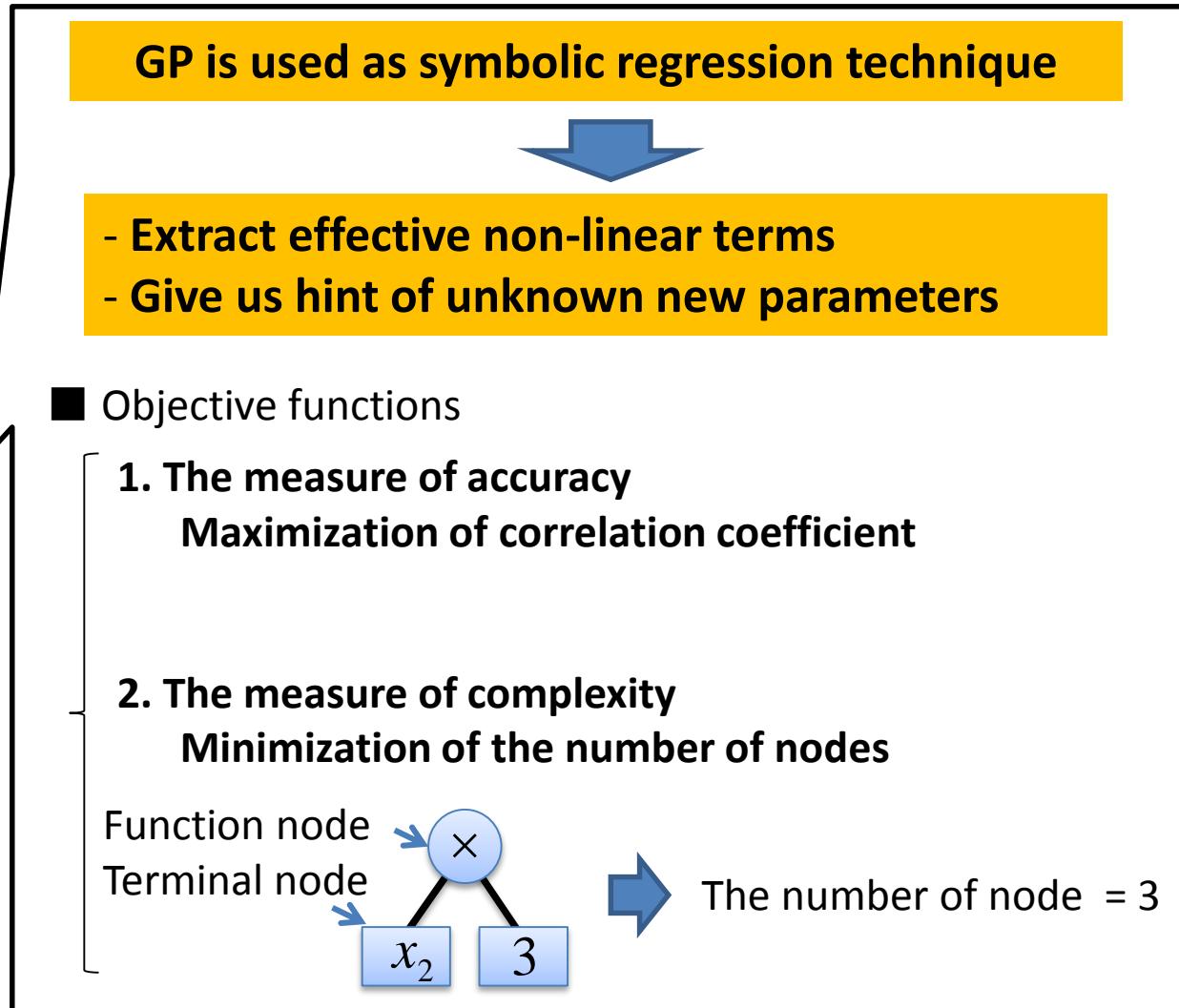
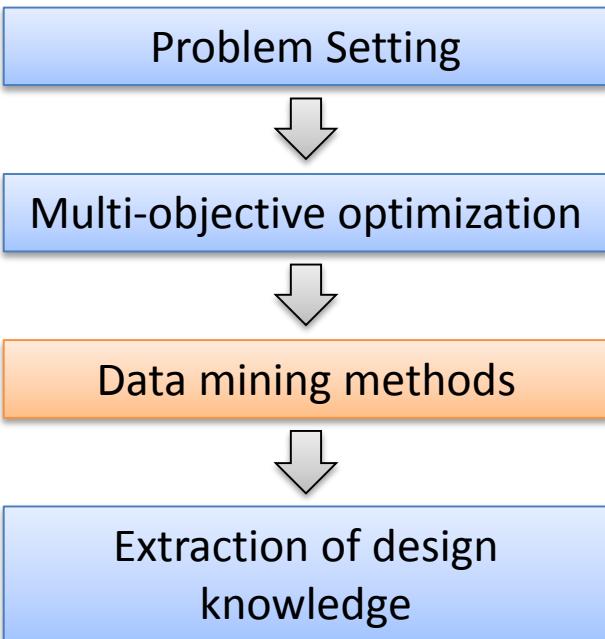
- Extension of Genetic Algorithm(GA) (Koza, 1990)
- Genome expression of GP is different from GA



- Genetic operators



# Genetic Programming in MODE



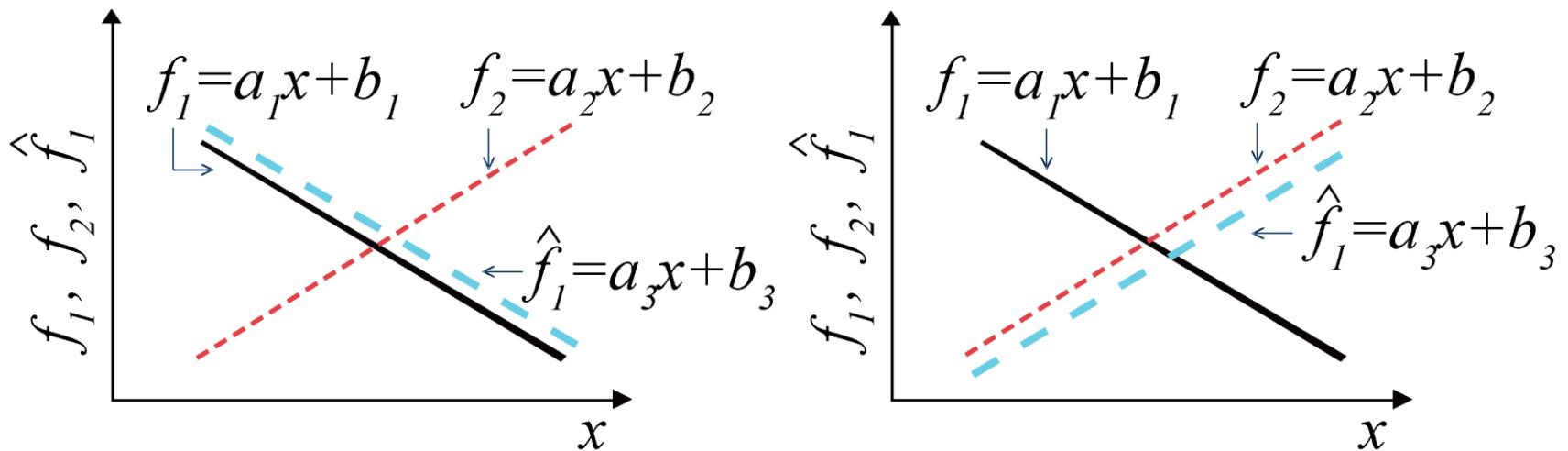
# Proposed MOGP

## - Measurement of Accuracy-

### Proposed GP

**Maximization of squared correlation coefficient**

$$Cor^2(f_i, \hat{f}) = \left( \frac{\sum_{j=1}^N (f_{i,j} - \bar{f}_i)(\hat{f}_j - \bar{f})}{\sqrt{(f_{i,j} - \bar{f}_i)^2} \sqrt{(\hat{f}_j - \bar{f})^2}} \right)^2 \longrightarrow \max \quad i = 1, 2, \dots, M$$



We want to extract “x” as design information

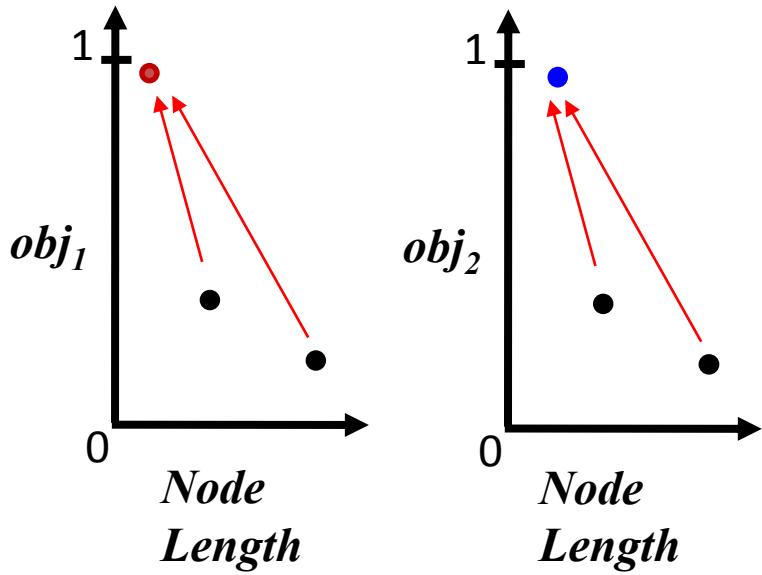
# Proposed MOGP

## - Objective Function -

**GP**

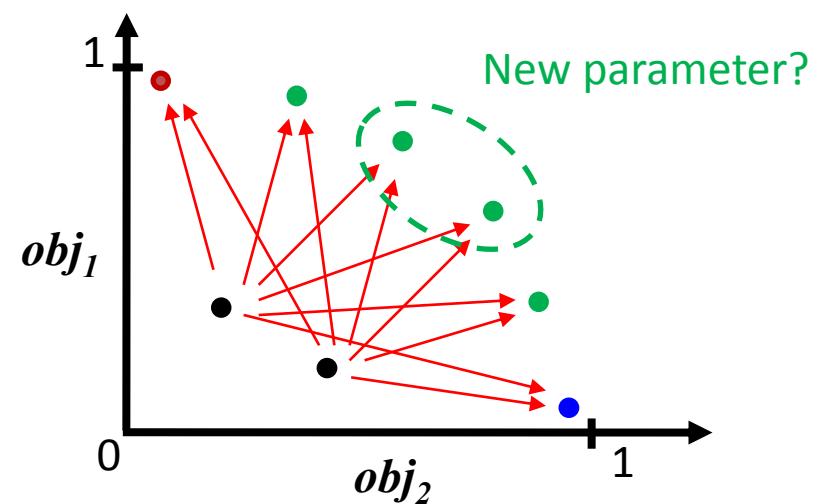
- 1) Maximization of correlation coefficient between  $obj_i$
- 2) Minimization of the number of nodes

$obj_i$ : *i*th objective function of MOO



**MOGP**

- 1) Maximization of squared correlation coefficient between  $obj_1$
- 2) Maximization of squared correlation coefficient between  $obj_2$
- ...
- N) Maximization of squared correlation coefficient between  $obj_N$
- N+1) Minimization of the number of nodes

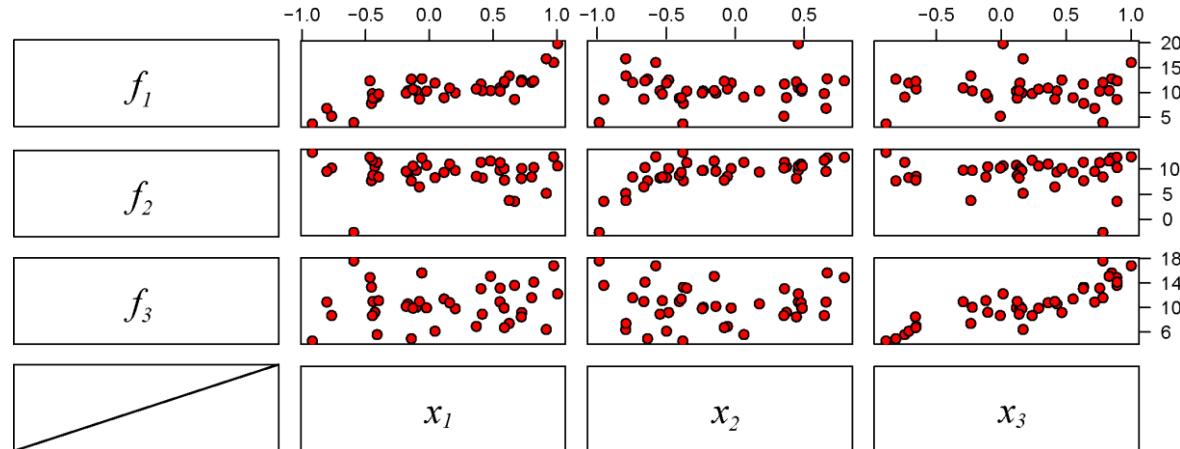


# Symbolic Regression Example

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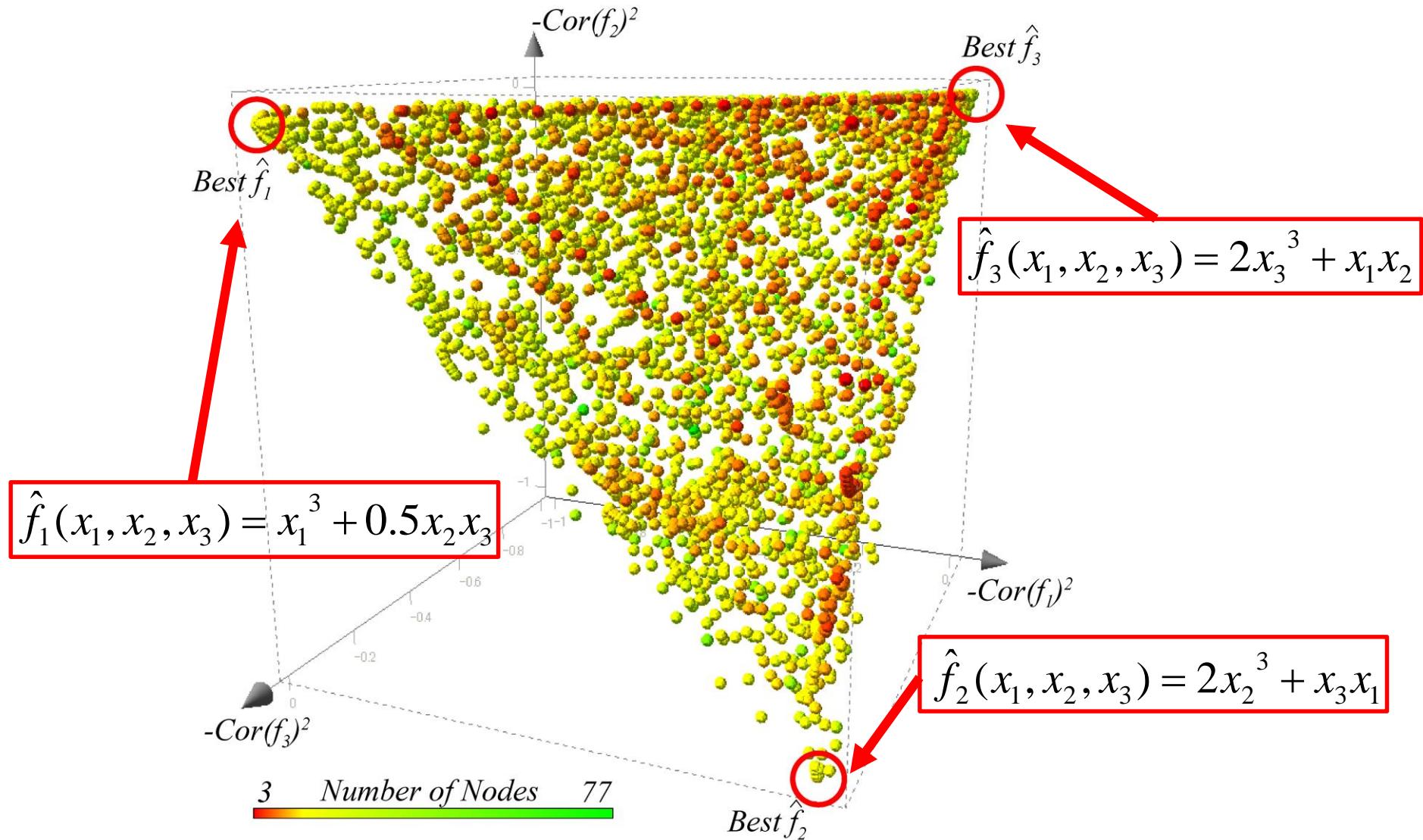
- Test function  $f_1(x_1, x_2, x_3) = 10x_1^3 + 5x_2x_3 + 10$
- $f_2(x_1, x_2, x_3) = 10x_2^3 + 5x_3x_1 + 10$
- $f_3(x_1, x_2, x_3) = 10x_3^3 + 5x_1x_2 + 10$        $x_1, x_2, x_3 = [-1, 1]$

- Data set
  - 40 sample points (random)



- GP
  - 1000 Individuals, 1000 Generations
  - 15 Trial

# Results of Proposed MOGP



# MODE of reusable launch vehicle (RLV)

## - Problem Setting -

### ■ Objective functions

#### Ascent phase

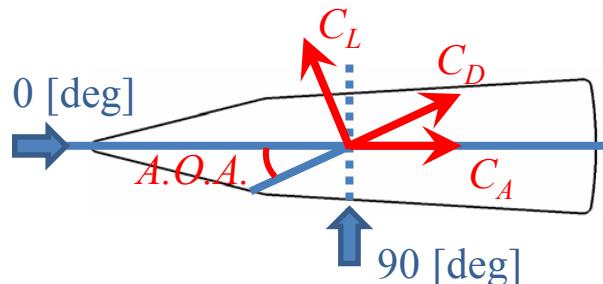
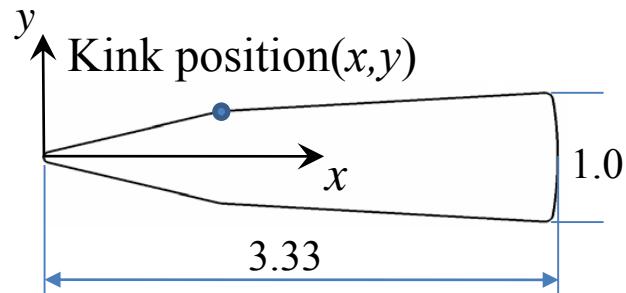
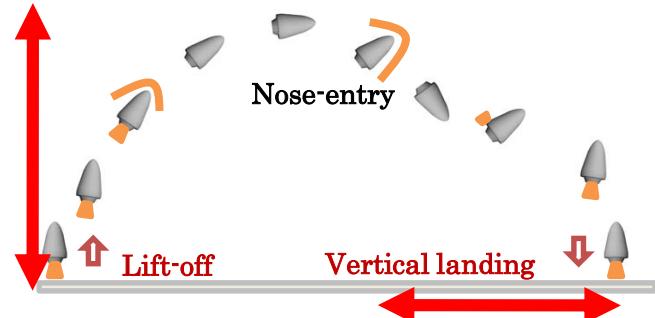
1. Drag to be minimized ( $M=2.0$ ,  $AOA = 0$ )

#### Return phase

2. Maximum L/D to be maximized at subsonic condition ( $M=0.8$ )
3. Maximum L/D to be maximized at supersonic condition ( $M=2.0$ )
4. Body volume to be maximized

### ■ Design parameters

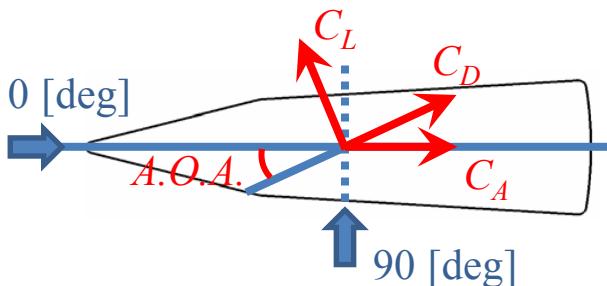
Kink position ( $x, y$ )



# Extraction of design information using Proposed MOGP

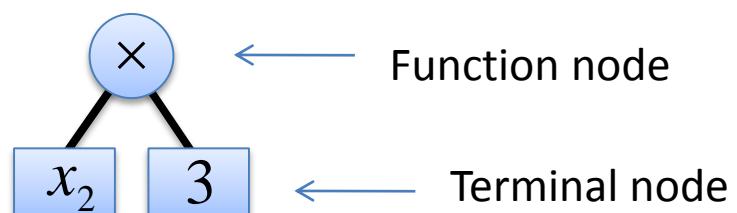
## Objective functions

1. Maximization of squared correlation between  $C_A, M=2.0, AOA=0$
2. Maximization of squared correlation between  $L/D_{MAX}, M=0.8$
3. Maximization of squared correlation between  $L/D_{MAX}, M=2.0$
4. Maximization of squared correlation between **Volume**
5. Minimization of the number of nodes in syntax tree

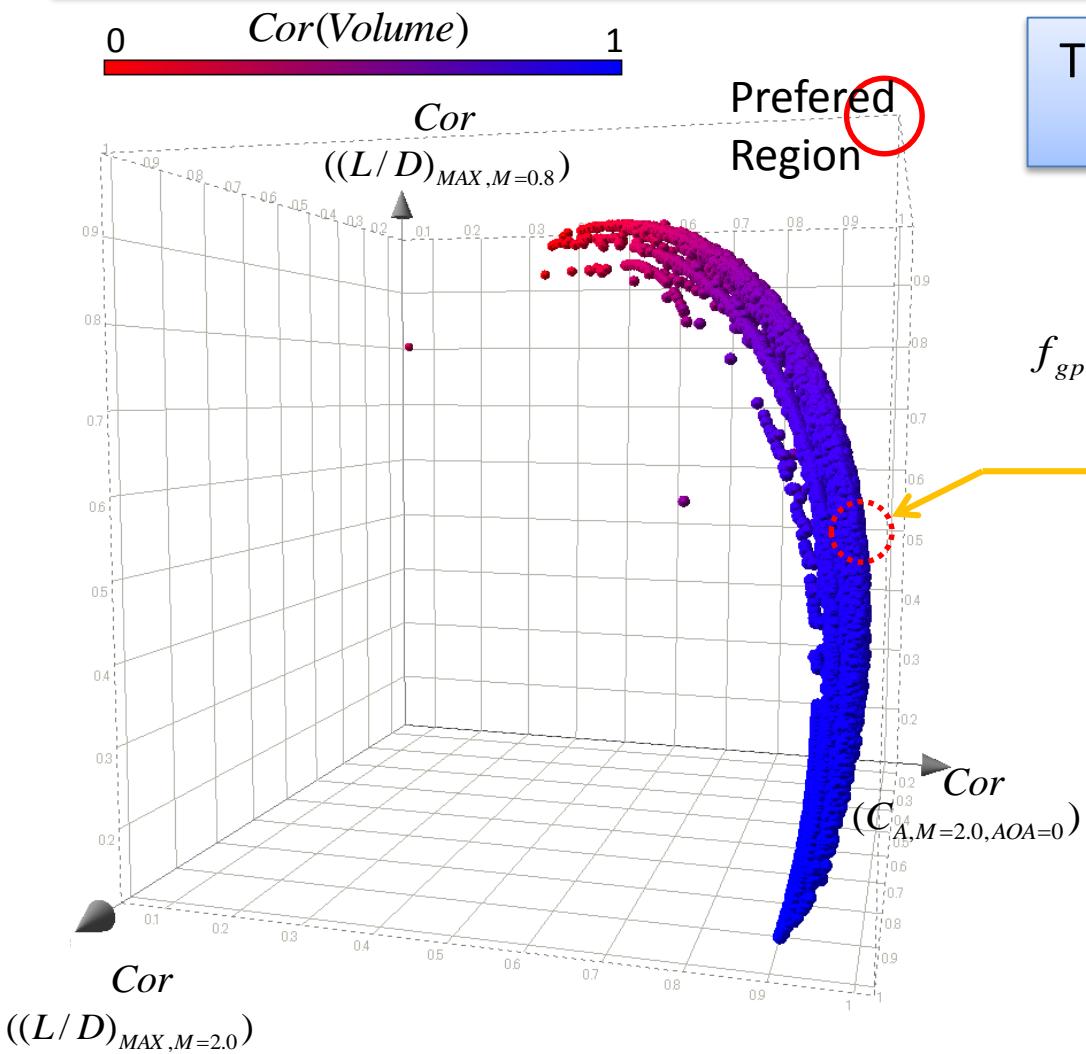


## Computational conditions

Gene type	Tree expression
Generation	1500
Population	1500
Crossover ratio	0.8
Mutation ratio	0.1
Function sets	+ , - , * , /
Terminal sets	Design parameter ( $x, y$ ), Constants [-1, 1]
Constraint	The number of nodes > 1



# Results of MOGP



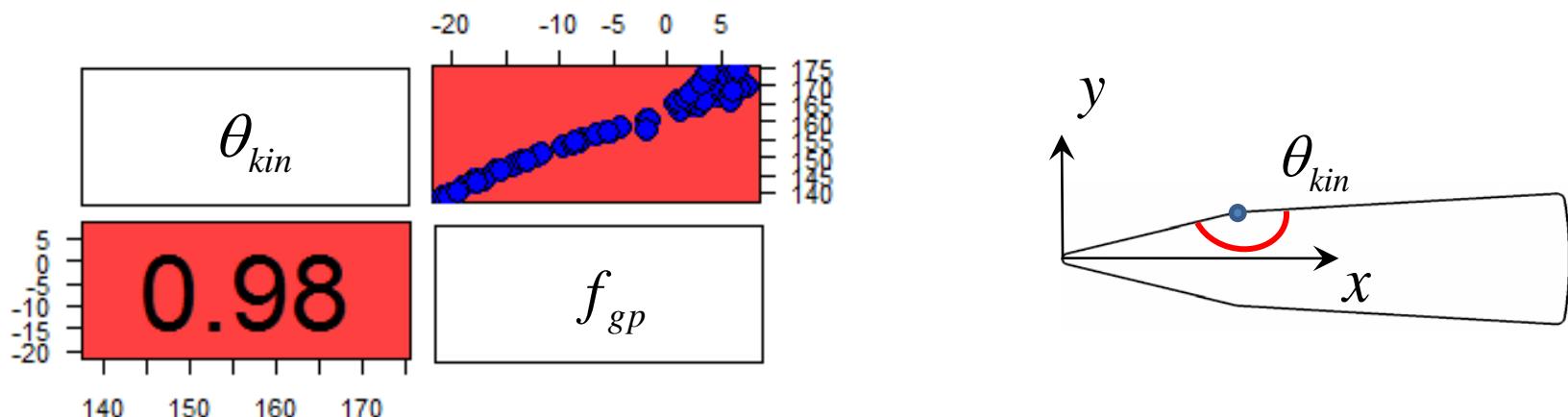
The maximum sum of all squared correlation

$$f_{gp} = -5.33y + 4.3xy + 4.3x - 1.28x^2 + xy^2 - 0.8x^2y - 0.33y^2$$

	Correlation
$C_{A,M=20,AOA=0}$	-0.99
$(L/D)_{MAX,M=0.8}$	0.44
$(L/D)_{MAX,M=2.0}$	0.98
<i>Volume</i>	-0.93

# Results of MOGP

$$f_{gp} = -5.33\underline{y} + 4.3\underline{xy} + 4.3\underline{x} - 1.28x^2 + \underline{xy^2} - 0.8x^2\underline{y} - 0.33\underline{y^2}$$



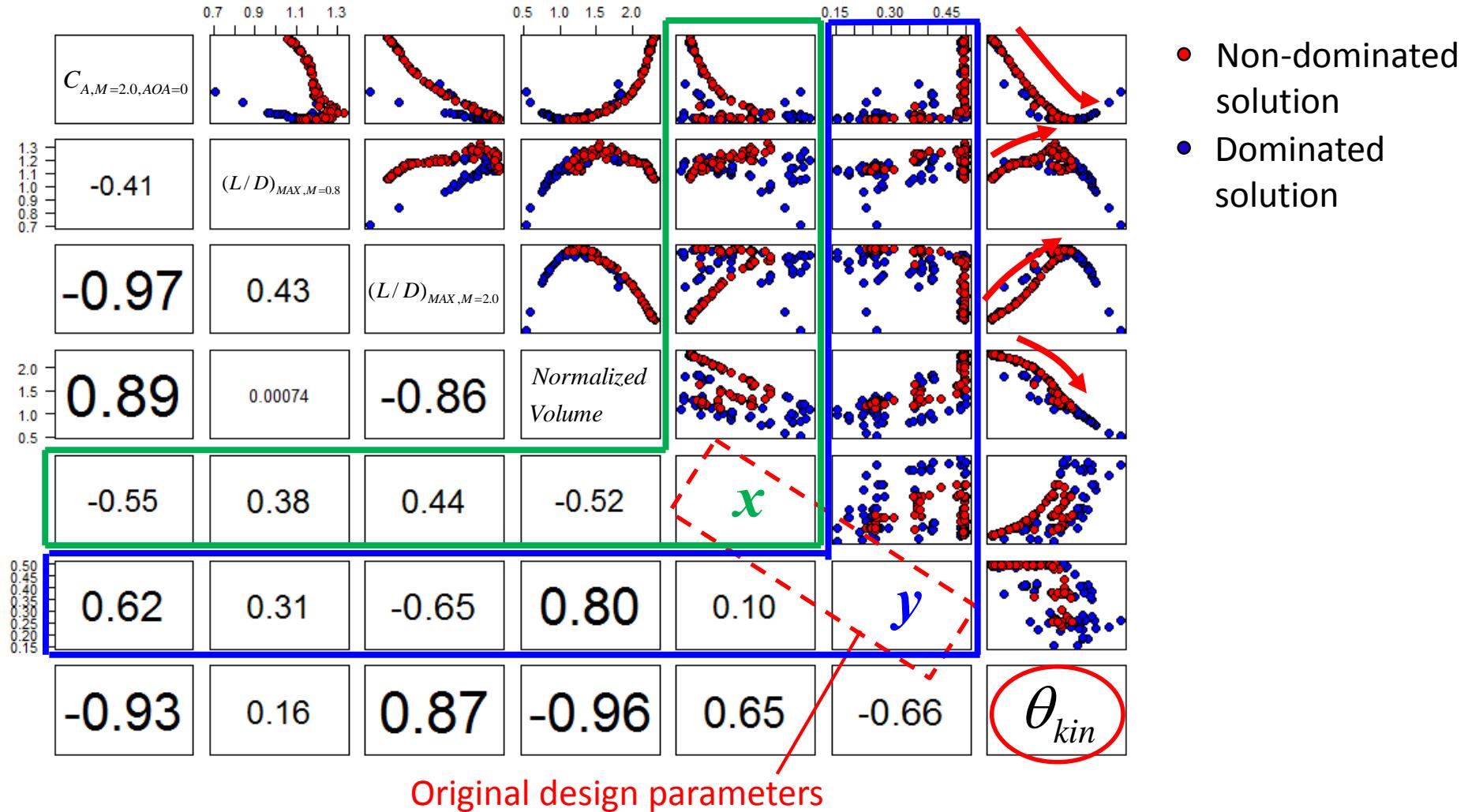
$$\theta_{kin} = 180 - \frac{180}{\pi} \arctan\left(\frac{y}{x}\right) + \frac{180}{\pi} \arctan\left(\frac{0.5-y}{3.33-x}\right)$$

$$= (188.53 - 16.8\underline{y} - 0.74\underline{y^2} + O[y^3])$$

$$+ \frac{180}{\pi y} + 2.52 - 4.83\underline{y} - 0.64\underline{y^2} + O[y^3] \underline{x}$$

$$+ \left(-\frac{60}{\pi y^3} + O[y^2]\right) + O[x^4] + (-1)^{\text{Floor}(\dots)} (-90 + O[y^3])$$

# Results of MOGP



# Conclusions

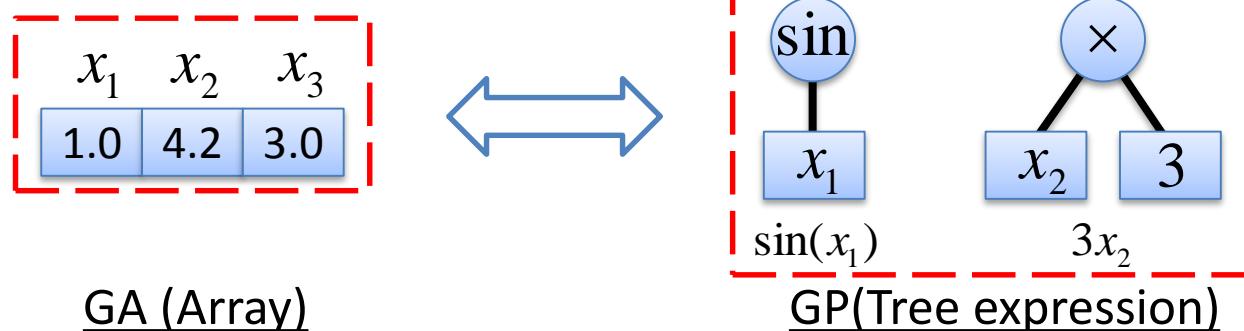
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- A new multi-objective genetic programming for design exploration is proposed.
  - The unique feature of MOGP is the simultaneous symbolic regression to multiple variables using correlation coefficients.
- MOGP is applied to non-dominated solutions of the design optimization problem of a bi-conical shape reusable launch vehicle.
  - The result of proposed MOGP presents symbolic equations which have high correlation to zero-lift drag at supersonic condition, maximum lift-to-drag at supersonic condition and volume of shape at the same time.
  - These results also have high correlation to kink angle of the body geometry. It implies that proposed GP is capable of finding composite new design parameters.

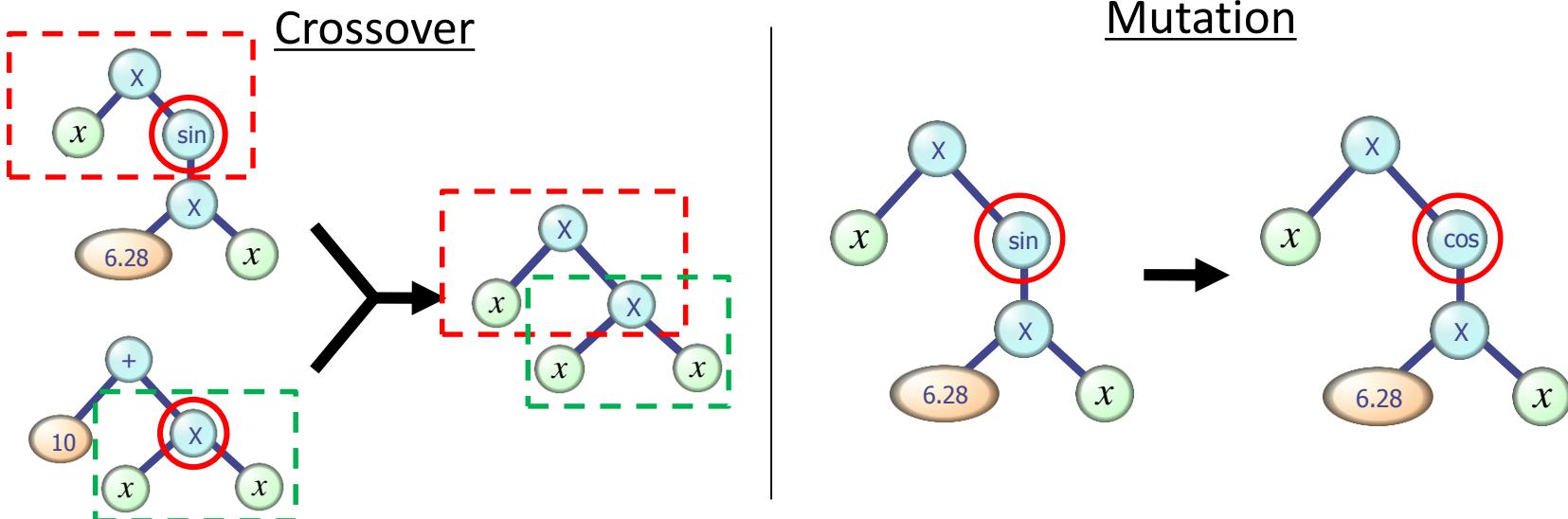


# About Genetic Programming

- Extension of Genetic Algorithm(GA) (Koza, 1990)
- Genome expression of GP is different from GA



- Genetic operators

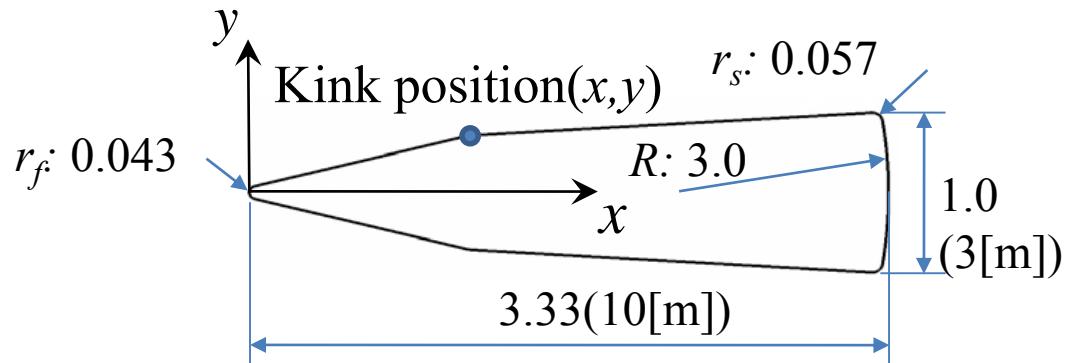


# MODE for RVT

## - Problem Definition -

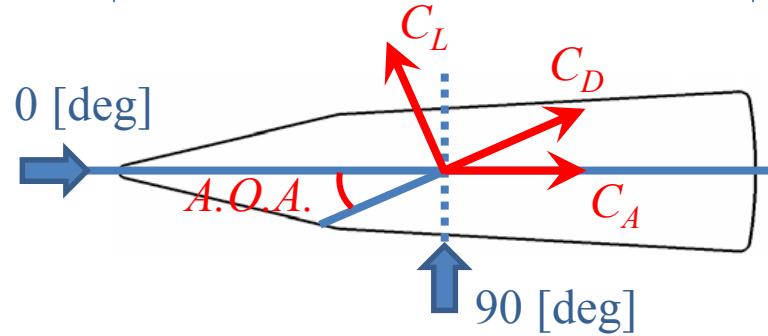
### ■ Design parameters

- Kink position( $x, y$ )



### ■ Constraint conditions

- Base diameter = 3[m]
- Length of the body = 10[m]
- Nose radius( $r_f$ ) = 0.043[m]
- Base corner radius( $r_s$ ) = 0.057[m]
- Base radius( $R$ ) = 3.0[m]



# Computational methods and conditions

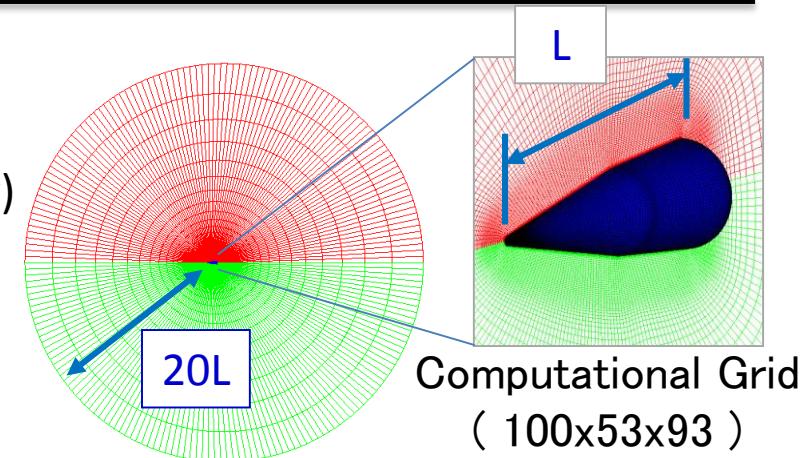
## ■ Computational Methods

### ■ CFD

- 3<sup>rd</sup> order MUSCL+SLAU (for shock instability)
- Baldwin-Lomax

### ■ MOEA

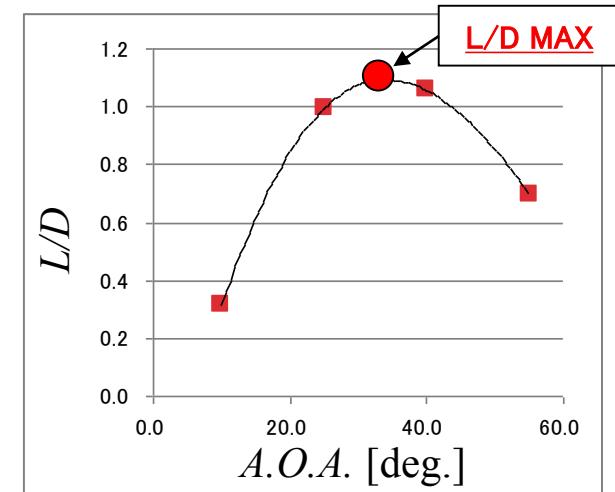
- Non-dominated Solution GA-II (NSGA-II)



## ■ Computational Conditions

### ■ Flow analysis

$(Re = 1.0 \times 10^7)$	Mach number	Angle of Attack [deg.]
Ascent phase	2.0	0
Return phase	2.0	10,25,40
	0.8	10,25,40,55

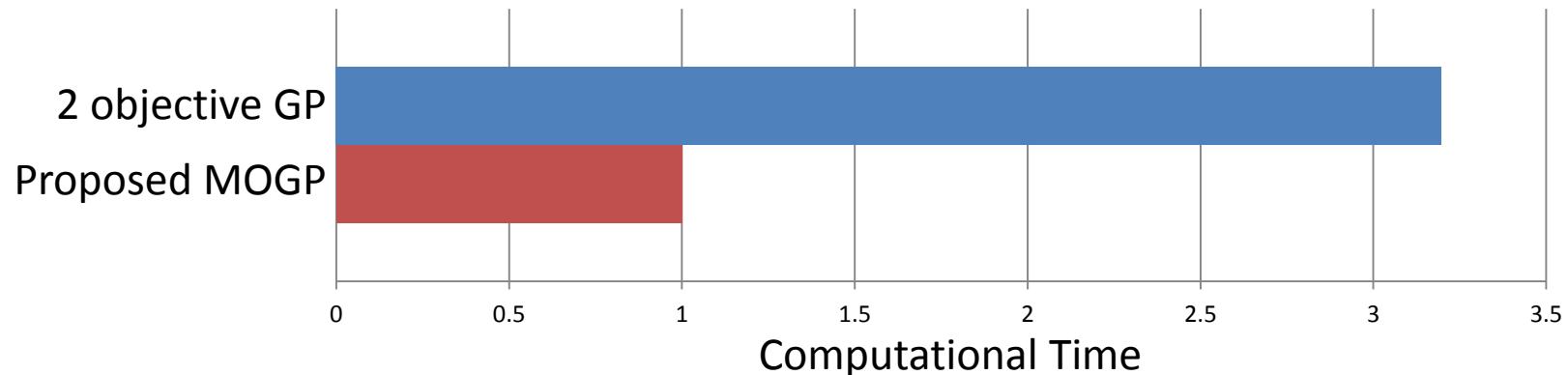


### ■ Optimization

- Population size: 20
- Generation size: 20 → 400 body shapes (3200 CFD runs) are evaluated

# Comparison of computational time

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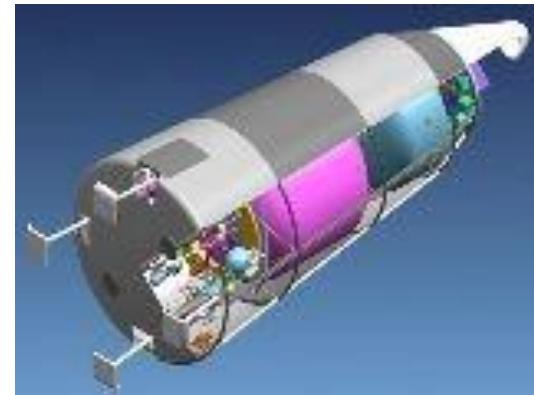
# Vertical Landing Reusable Launch Vehicle (RLV)

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- One of the future space transportation systems
- Motivation of development
  - Low cost, reusability and reliability



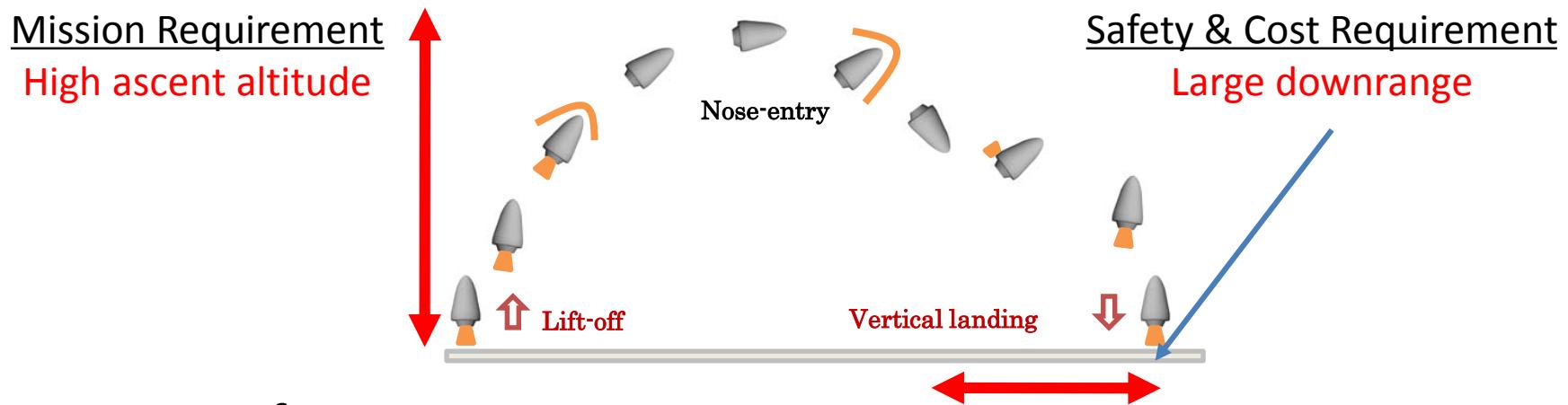
Reusable Vehicle Testing(RVT)



Reusable Sounding  
Rocket Vehicle(RSRV)

# Vertical Landing Reusable Launch Vehicle (RLV)

## ■ Flight sequence & Design requirements



## ■ Important feature

- RLV is a vertical landing SSTO rocket.
- RLV flies over a wide range of the flight speed and attack angles.

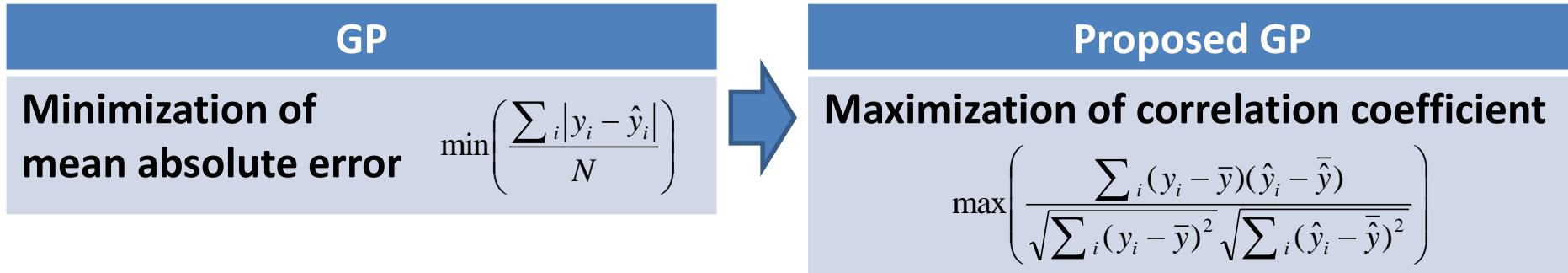
It is important to understand relationship  
between shape parameter and flow field



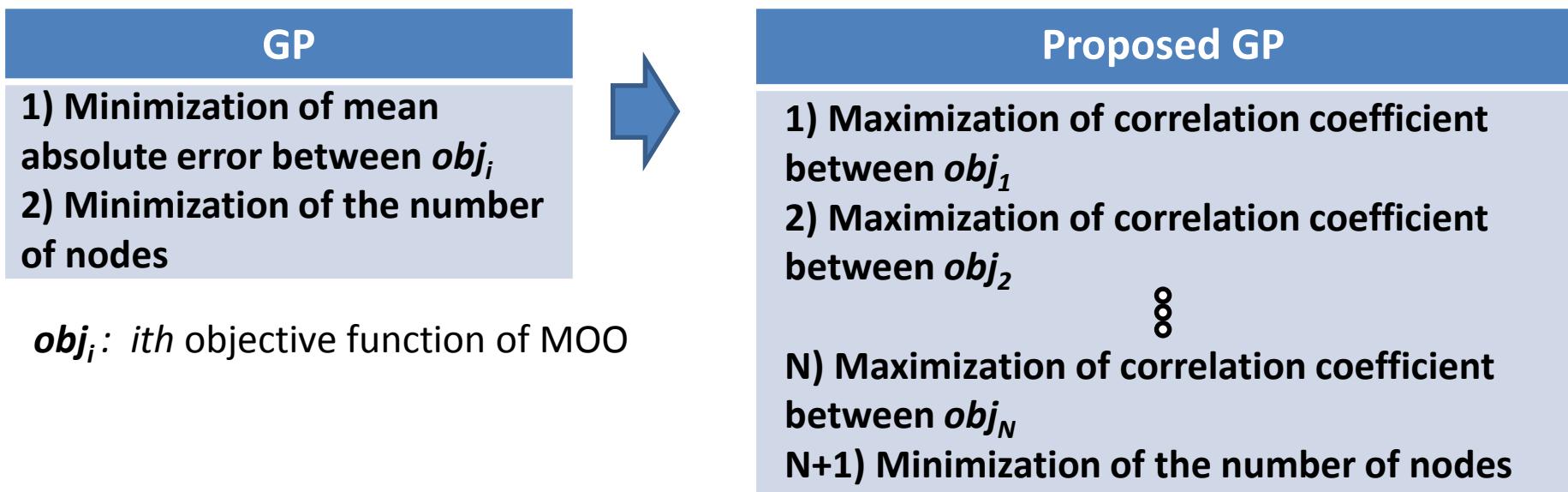
More knowledge on aerodynamic shape design is necessary

# A new type of Genetic Programming

## 1. Change the measure of accuracy

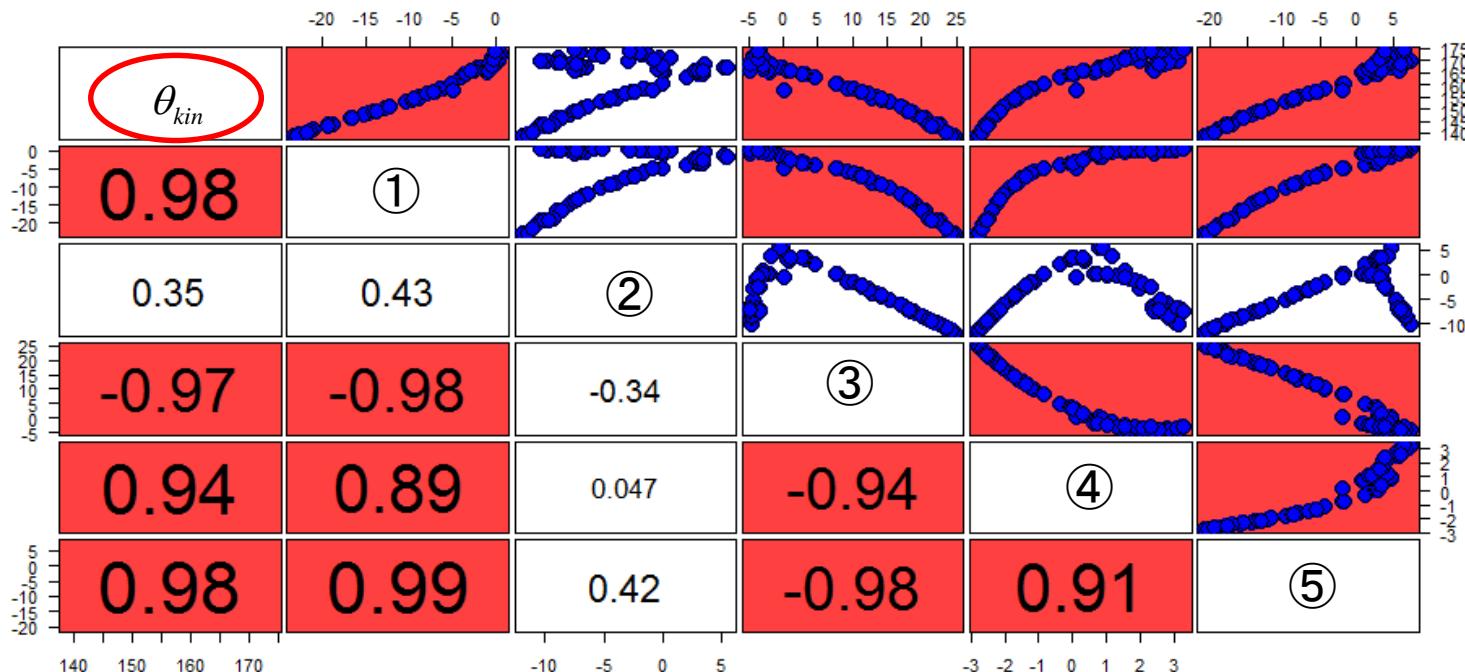


## 2. Change the number of objective functions



# Results of proposed MOGP

		$x$	$y$	$x^2$	$y^2$	$xy$	$xy^2$	$x^2y$	$x^3$	$Const.$
①	Maximum $\text{Cor}(C_{A,M=2.0,AOA=0})$	2.75	-4.1	-2.6	-2	3.1		-1	1	-1.65
②	Maximum $\text{Cor}(L/D_{MAX,M=0.8})$	0.7	0.6	-1	-4	4	2			0.4
③	Maximum $\text{Cor}(L/D_{MAX,M=2.0})$	-3.87	8.75	1.3	5	-0.1				-0.3
④	Maximum $\text{Cor}(Volume)$	1	-2		0.2	0.1				0.3
⑤	Maximum sum of all correlation	4.3	-5.33	-1.28	-0.33	4.3	-0.8			



# Multi-Objective Design Exploration

## - Problem Setting -

Problem Setting



Multi-objective optimization



Data mining methods



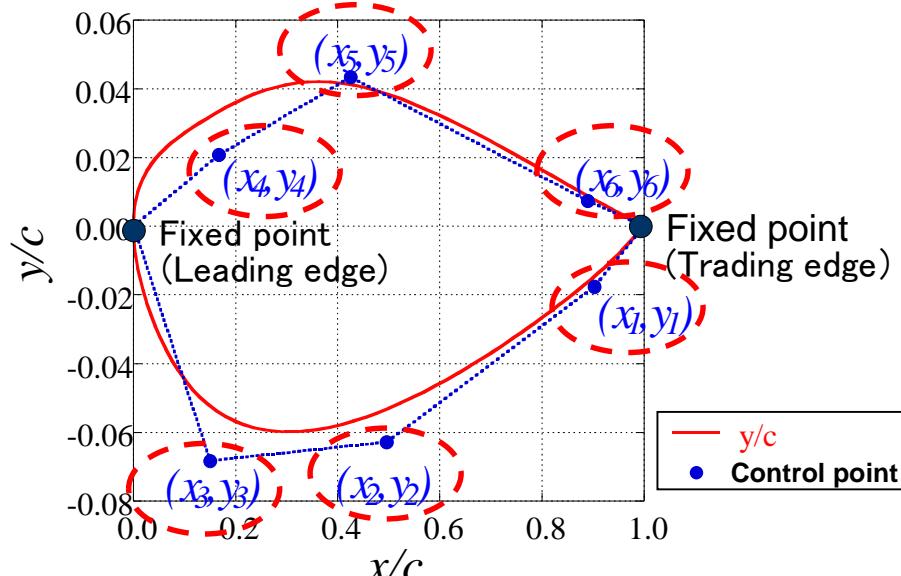
Extraction of design knowledge

e.g.) MODE for 2D wing shape

### ■ Objective functions

1. Maximization of lift coefficient( $C_L$ )
2. Minimization of drag coefficient( $C_D$ )

### ■ Design parameters



# Multi-Objective Design Exploration

- Multi-objective optimization -

Problem Setting



Multi-objective optimization

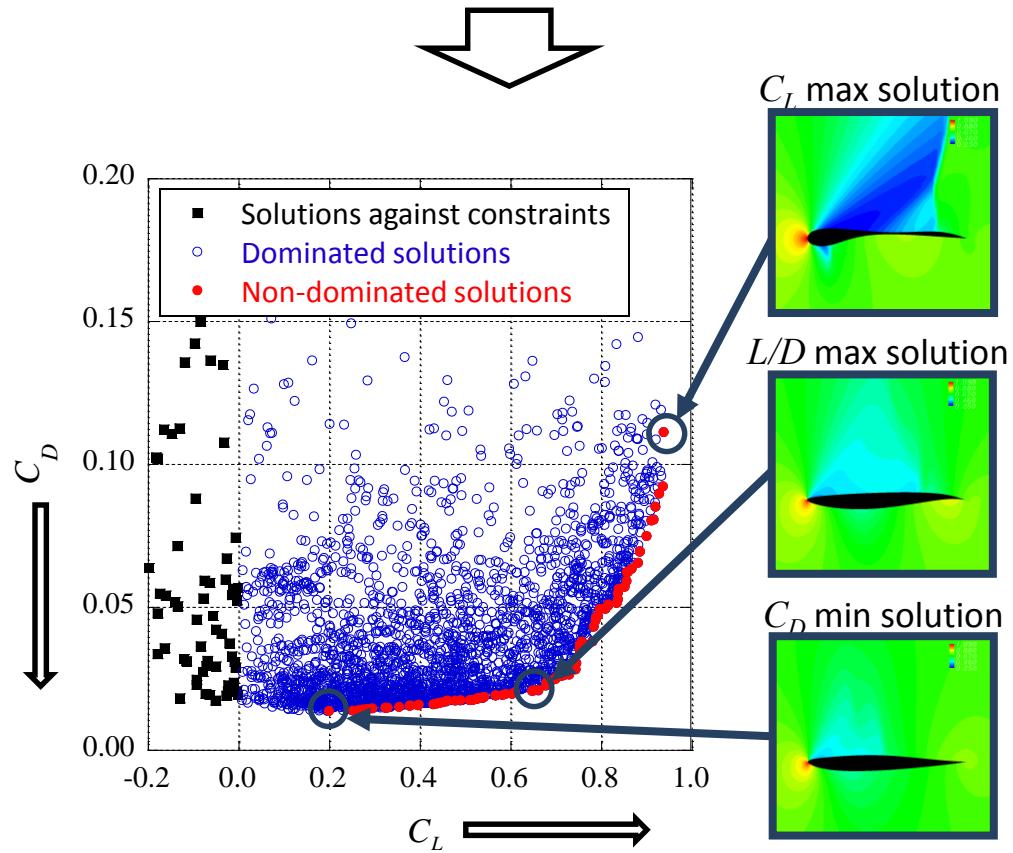


Data mining methods



Extraction of design  
knowledge

**Non-dominated solutions** are obtained by using multi-objective evolutionary algorithms



# Multi-Objective Design Exploration

- Data mining methods -

Problem Setting



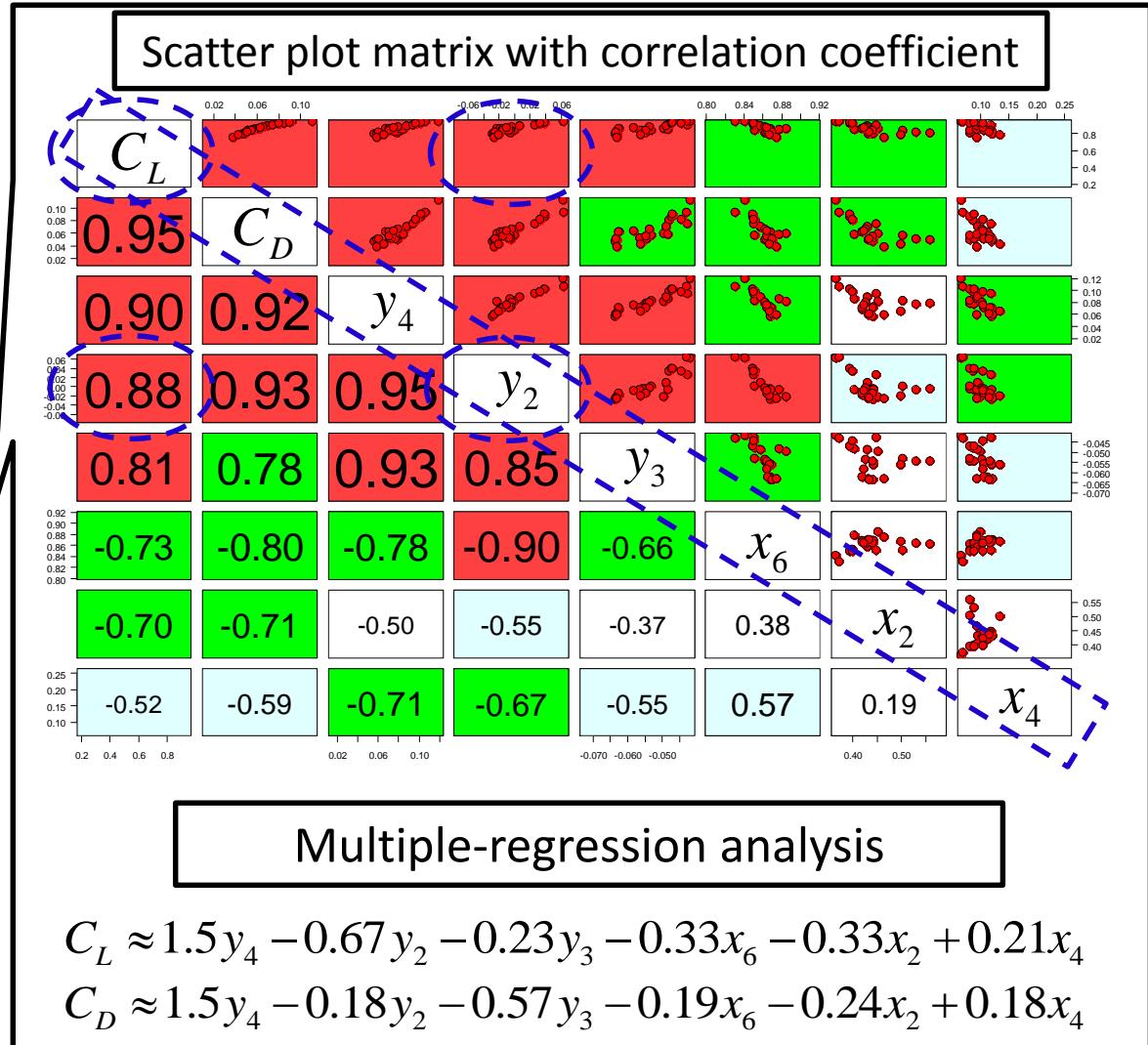
Multi-objective optimization



Data mining methods



Extraction of design  
knowledge



# Computational methods and conditions

## ■ Computational Methods

### ■ CFD

- 3<sup>rd</sup> order MUSCL+SLAU (for shock instability)
- Baldwin-Lomax

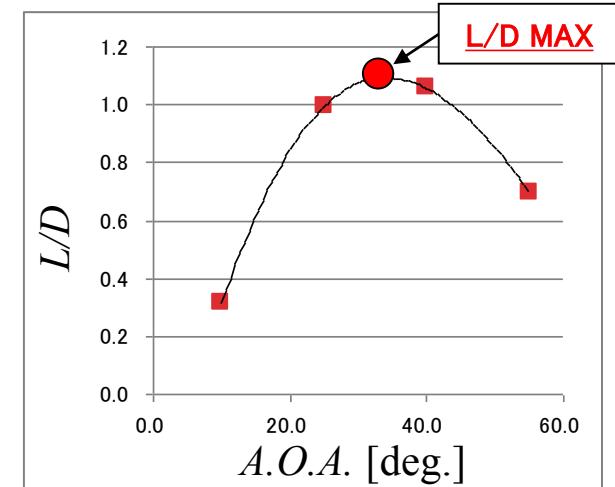
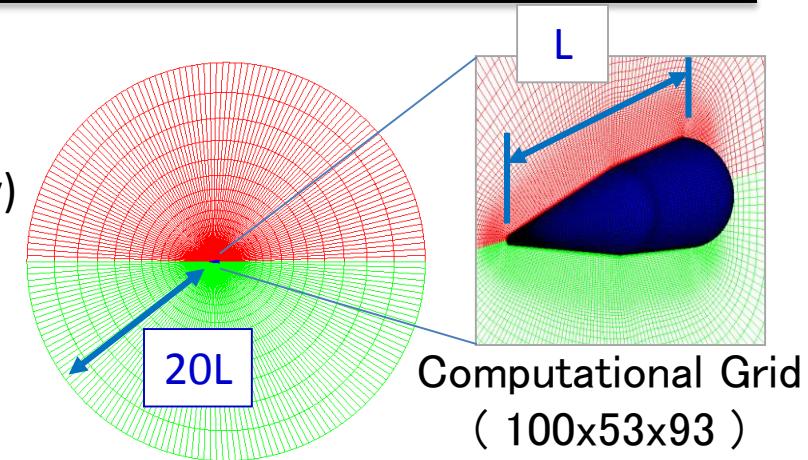
### ■ MOEA

- Non-dominated Solution GA-II (NSGA-II)

## ■ Computational Conditions

### ■ Flow analysis

$(Re = 1.0 \times 10^7)$	Mach number	Angle of Attack [deg.]
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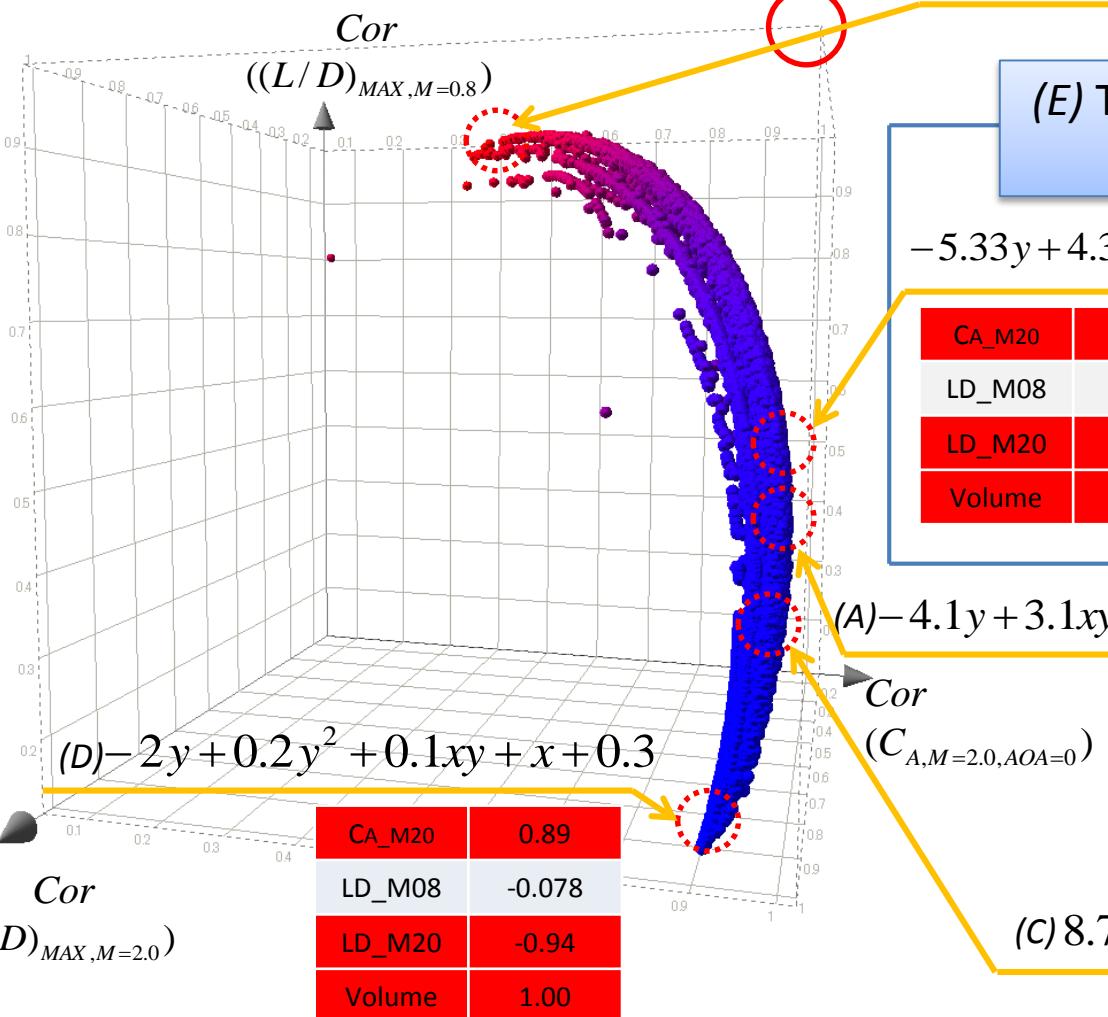


### ■ Optimization

- Population size: 20
- Generation size: 20 → 400 body shapes (3200 CFD runs) are evaluated

# Results of MOGP

CA_M20	-0.44
LD_M08	0.97
LD_M20	0.35
Volume	-0.11



$$(B) -4y^2 + 4xy + 2xy^2 - x^2 + 0.7x + 0.6y + 0.4$$

(E) The maximum sum value of all squared correlation

$$-5.33y + 4.3xy + 4.3x - 1.28x^2 + xy^2 - 0.8x^2y - 0.33y^2$$

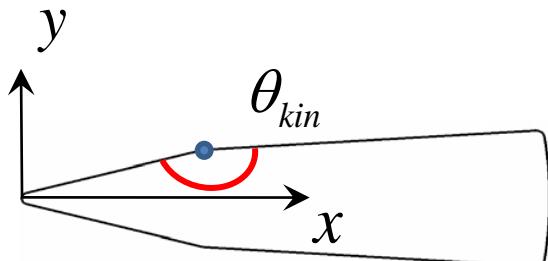
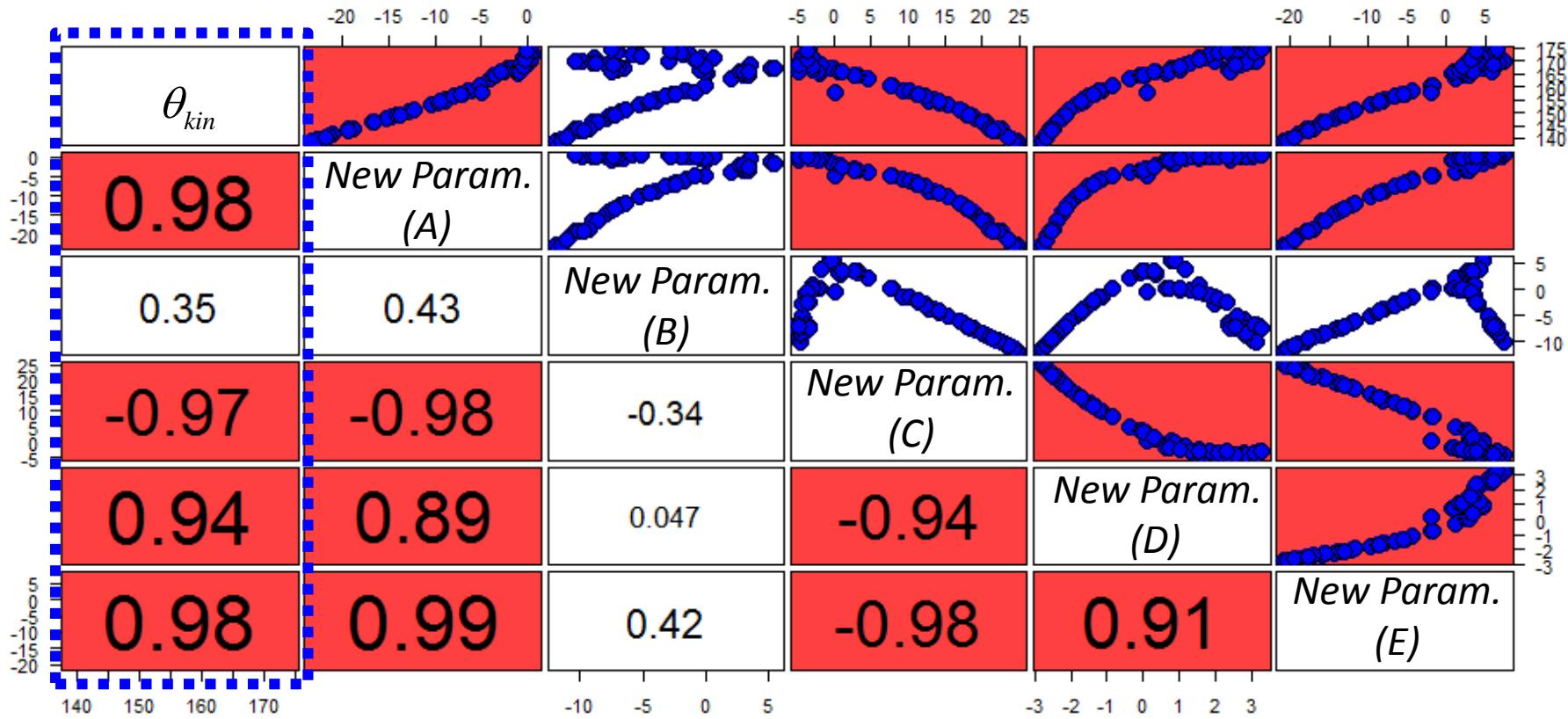
CA_M20	-0.99
LD_M08	0.44
LD_M20	0.98
Volume	-0.93

CA_M20	-1.0
LD_M08	0.46
LD_M20	0.98
Volume	-0.91

$$(A) -4.1y + 3.1xy + 2.75x - 2.6x^2 - 2y^2 - x^2y + x^3 - 1.65$$

CA_M20	0.98
LD_M08-	-0.36
LD_M20	-1.0
Volume	0.96

# Results of MOGP



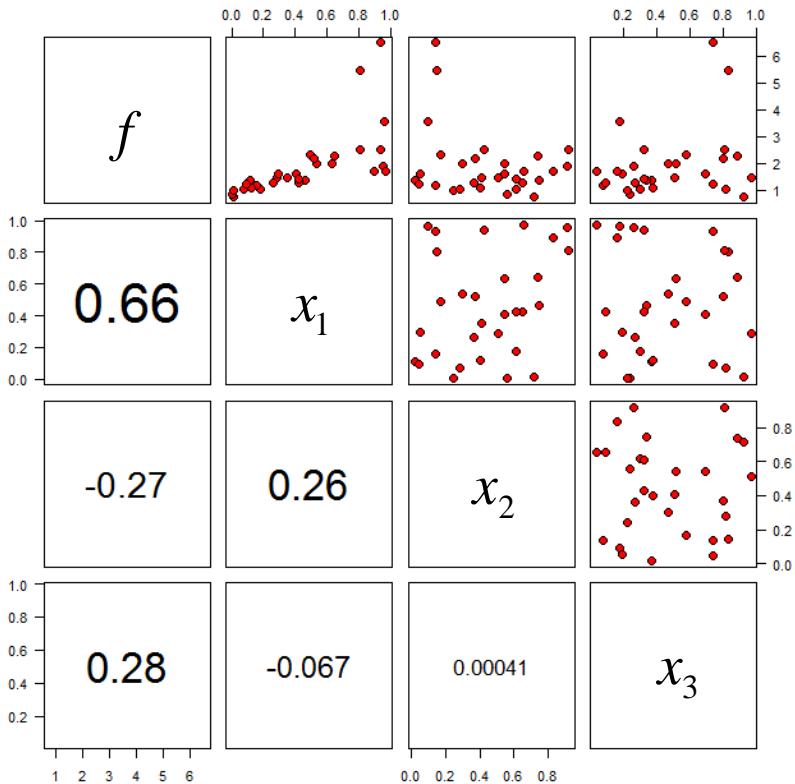
$$\theta_{kin} = 180 - \frac{180}{\pi} \arctan\left(\frac{y}{x}\right) + \frac{180}{\pi} \arctan\left(\frac{0.5-y}{3.33-x}\right)$$

# Example of GP symbolic regression

Test function

$$f(x_1, x_2, x_3) = \sin(x_1) + \cos(x_2) + \frac{x_1^2 x_3}{x_2} + x_1 x_2 x_3$$

Data set

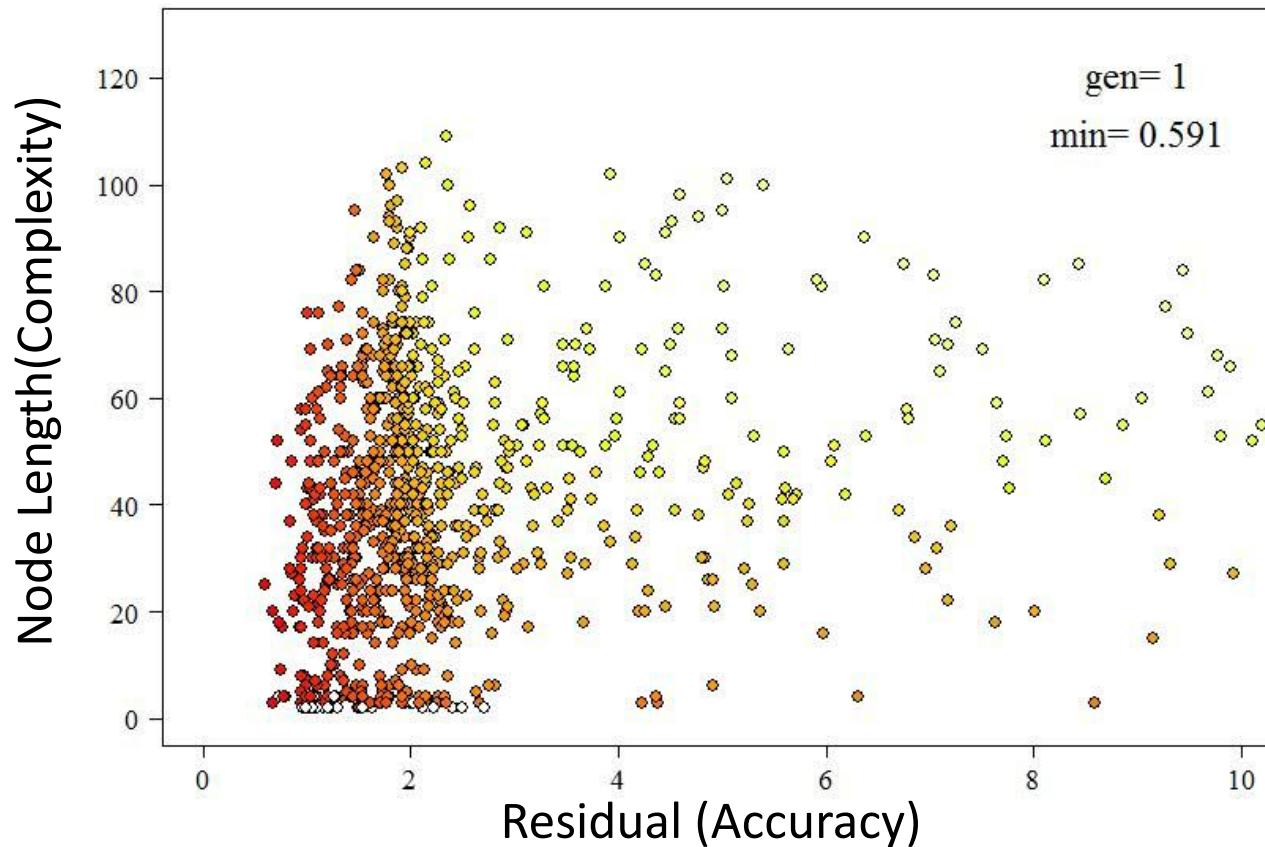


Computational Condition

Gene type	Tree expression
Generation	1000
Population	1000
Crossover ratio	0.8
Mutation ratio	0.2
Function sets	+, -, *, /, sin, cos
Terminal sets	Design parameters, Constants [-1, 1]
Constraint	The number of nodes > 1

# Example of GP

## - Result -



574<sup>th</sup> generation →  $f(x_1, x_2, x_3) = \sin(x_1) + \cos(x_2) + \frac{x_1^2 x_3}{x_2} + x_1 x_2 x_3$

# symbolic regression example

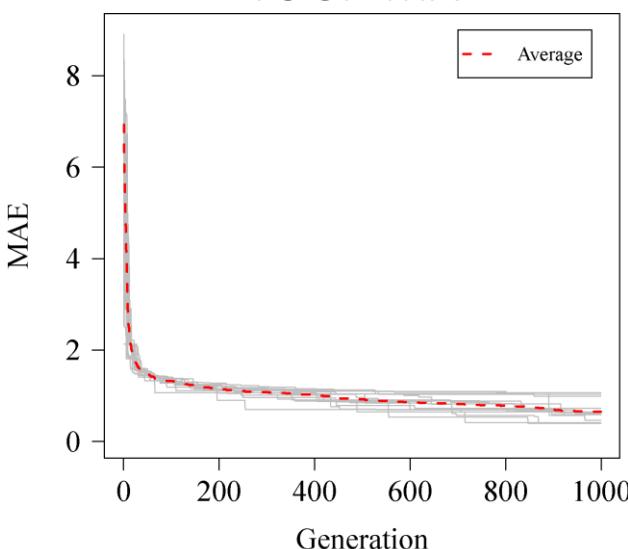
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- Test function  $f_1(x_1, x_2, x_3) = 10x_1^3 + 5x_2x_3 + 10$   
 $f_2(x_1, x_2, x_3) = 10x_2^3 + 5x_3x_1 + 10$   
 $f_3(x_1, x_2, x_3) = 10x_3^3 + 5x_1x_2 + 10 \quad x_1, x_2, x_3 = [-1,1]$
- Data set
  - 40 sample points (random)
- GP
  - 3 type of GP
    - TOGP-MAE (2 Objective GP, Mean Absolute Error)
    - TOGP-SCC (2 Objective GP, Squared Correlation Coefficient)
    - MOGP-SCC (Multi Objective GP, Squared Correlation Coefficient)
  - 1000 Individuals, 1000 Generations
  - 15 Trial

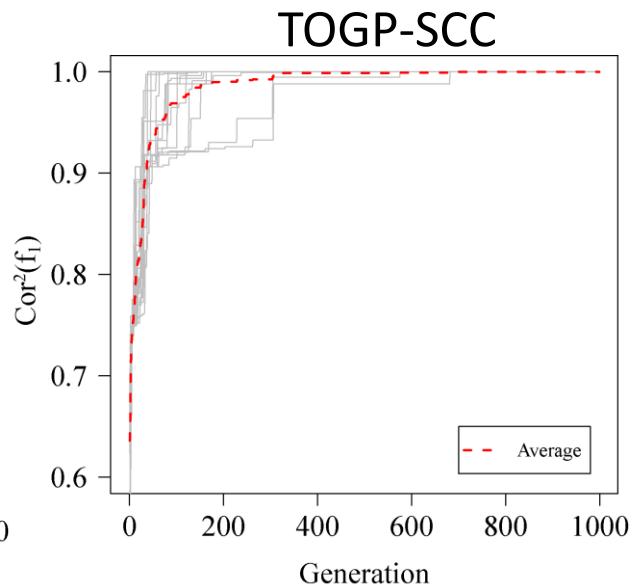
# Results

## - History of accuracy

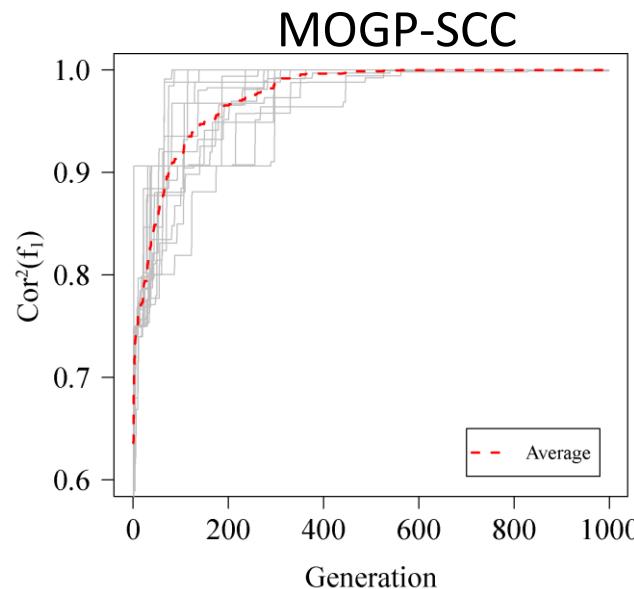
TOGP-MAE



TOGP-SCC



MOGP-SCC



## - Comparison of computational time

